

Jerk

AUTONOMOUS FLIGHT TERMINATION UNIT (AFTU)

AFTU Inputs (3) RS-422 inputs for sensor data (GPS, IMU) (2) MSTR ARM/SAFE CMD (A & B) (1) INT PWR CMD (1) EXT PWR CMD (2) Discrete liftoff indicators (A & B) (2) Ethernet IP address selects (1) 10 W, 28 VDC output enable (1) Battery voltage and temperature monitor (2) GPS RF Outputs (2) Destruct warning TLM outputs (4) Destruct 5.25 A min outputs (current limited) (1) RS-422 output for AFTU TLM PCM data (1) Resistor shunt current monitor (1) GPS output (for cross-strap to second AFTU) (1) +5 V, 1 A current limited output for ext. GPS power (1) +15 V, -15 V, 0.15 A current limited output (2) 10 W, 28 VDC outputs I/O (1) Full-duplex RS-422 interface for AFTU cross strap (2) Ethernet interfaces for command and status (2) RS-422 interfaces for command and status **Power Supply** Supply Voltage +28 V primary power, +/-15 V, +5 V power outputs **Power Consumption** < 25 W (including GPS) **Embedded GPS Receiver** Supported Signals M-Code, SAASM P(Y), C/A Data Output Rate 10 Hz Position Accuracy $22\,m\,3\,\sigma$ Velocity Accuracy .9 m/s 3 σ 196 m/s² Acceleration Max

108 m/s³



The L3Harris Autonomous
Flight Termination Unit (AFTU)
is a configurable, fault-tolerant
autonomous flight termination unit
and the core of L3Harris' autonomous
flight safety system. The only RCC319-certified flight termination unit
with M-code GPS compatibility, the
AFTU is poised to support present
and future launch requirements into
2030 and beyond.

The AFTU integrates multiple functions into a single unit to keep the system affordable, compact and reliable:

- > CASS processor
- > Power changeover switch
- > Redundant initiation circuits
- > RCC-324-compliant M-code GPS receiver (option)

Built on a 50-year history of flight termination experience, the AFTU leverages heritage circuit designs on all functions listed above, with proven performance on Atlas V, Delta IV, Space Launch System, Space Shuttle, ISS and many other missions. The AFTU is single-fault tolerant, with redundancy in the master safe/arm, local arm, liftoff indicators, initiation circuits and all flight-critical FPGAs. Local and remote cross-strap heartbeat monitors ensure safe utilization of redundant hardware configurations. The embedded power changeover switch can disable the AFTU once out of the range-safety boundary for further fault protection.

L3Harris has been providing rock-solid flight termination hardware for over 50 years with zero operational failures across all product lines. The AFTU can be trusted to perform flawlessly on every mission.

PHYSICAL AND ENVIRONMENTAL CHARACTERISTICS	
Physical	
Volume	122.5 in³
Dimensions	2.5" H x 7" W x 7" D
Weight	4.5 lb
Reliability	
Operating Life	10,000 hours
Storage	15 years
Reliability	> 0.9999 at 95% confidence
Environments (Qual)	
Thermal environment (TBD)	-37 °C to +71 °C
Pyro Shock	> 5600 G @ 10,000 Hz
Acceleration	20 g 30 sec ea ± axis (180 sec total)
Random Vibration	36 Grms, 300 sec/axis (test 1) 58 Grms, 60 sec/axis (test 2)

BUILT-IN M-CODE GPS RECEIVER

The AFTU includes an optional embedded GPS receiver based on the L3Harris IEC M2 GRAM Type II, the only M-code GPS receiver qualified for use by the GPS Directorate. In addition to cost, weight and space savings, the embedded GPS provides built-in compliance to the congressional M-code mandate, saving future time and expense.

EXPERT SUPPORT

The AFTU is designed, built, assembled and tested all within one facility and is serviced and supported by engineering professionals with decades of spaceflight design experience. Every AFTU delivered is accompanied by domain expertise in parts, materials, radiation analysis, mechanical engineering, power supply design, digital signal processing, radio frequency design and manufacturing engineering. For most applications, existing data items can be provided for review, reducing the analysis and testing required.

Autonomous Flight Termination Unit (AFTU)

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1025 W. NASA Boulevard Melbourne, FL 32919 t 800 852-5105 f 513 573-6290 SpaceSales.cin@L3Harris.com